

Mars Today 1

An immediate and inexpensive
program for manned Mars visitation

Basic Assumptions

- No nuclear components
- No heavy lift launch capability
- No oversized components
- Use existing components to minimize development
- Emergency supplies for crew
- Recycle ?

Major systems

- Space Propulsion
- Transit Habitation
- Surface Habitation
- Mars Lander
- Crew Ferry Vehicle(s)

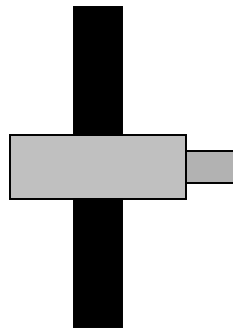
Space Propulsion



- Centaur (Mentaur)
- Empty Mass 2200Kg
- Lox/H2 16,800Kg
- Lox/CH4 42,200kg

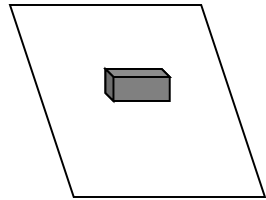
A Mentaur is a Centaur upper Stage with modified RL-10 Engines and propellant tanks for CH4 instead of LH2. Same shape, different mass.

Transit Habitation



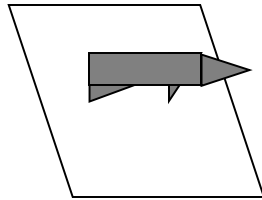
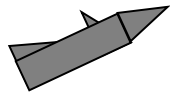
- ‘ The Mir core’
- 20,000 Kg
- Orbital crew habitat
- Long range Earth communications
- 16000 Kg Provisions

Surface Habitation Equipment



- Multiple deliveries
- 3m diameter
- 5m long
- 10,000Kg

Mars Lander



- 20,000 kg GLOW
- Hypersonic
aerodynamic L/D 1.0
- 10,000 Kg Payload
(surface module etc)
- Mars launch capability
(if resupplied with
10,000 kg propellant)

Crew Ferry Vehicle

- Deliver and retrieve crew
- Space Shuttle
- Soyuz/Apollo/etc

Mars Today Supply Profile

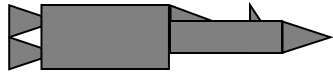
- LEO Assembly
- First LEO Impulse
- Trans Mars Impulse
- Trans Mars Cruise
- Direct Entry to Landing

Supply First LEO Impulse



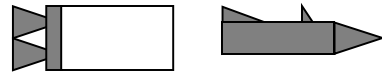
- Orbit = 300x300 km Alt
- Mass 58 (116) Mg
- Burn Centaur (Mentaur)
- $\Delta V = 1.5$ km/s
- New Orbit =
300x11000km Alt
- Stage Centaur

Supply Trans Mars Impulse



- Mass 39 (72) Mg
- Burn Most of Centaur
- $\Delta V = 2.4$ km/s
- Escape Excess $V = 4.3$ km/s or less

Supply Cruise / Entry

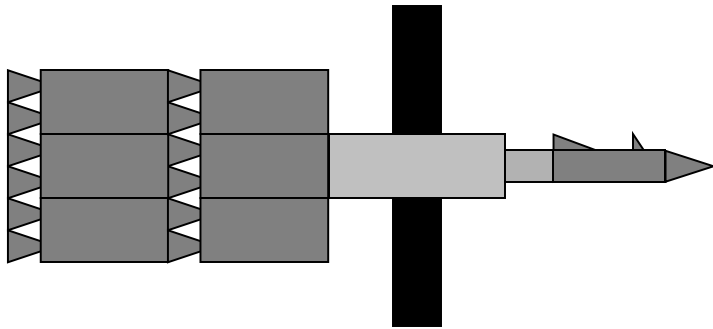


- Time 9 months
- Available DeltaV = 0.25 km/s
- Aerobrake for Mars Orbit Capture
- Autoland / Remote

Mars Today Manned Profile

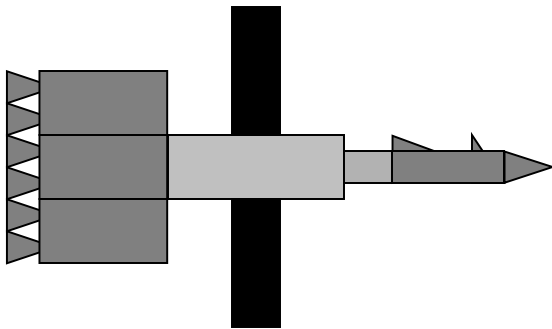
- LEO Assembly
- First LEO Impulse
- Trans Mars Impulse
- Trans Mars Cruise
- Mars Orbit Capture
- Mars Braking Descent
- Mars Surface Stay
- Mars Orbit Rndvs
- Mars Lander Return
- Trans Earth Impulse
- Earth Orbit Capture
- Crew Recovery

Manned First LEO Impulse



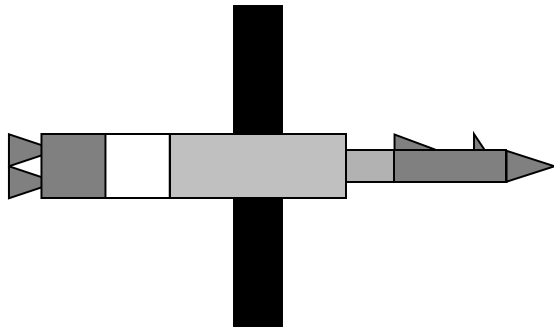
- Burn Four Centaurs
- $\Delta V = 1.7 \text{ km/s}$
- New Orbit =
300x1300 km Alt
- Stage Centaurs

Manned Trans Mars Impulse



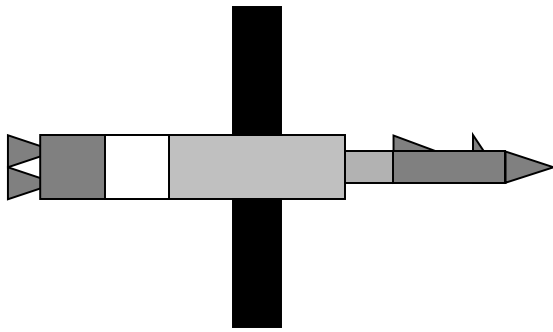
- Burn 3.25 of Centaurs
- $\Delta V = 4.2$ km
- $C3 = 4.3$ km or less depending on req.

Manned Cruise



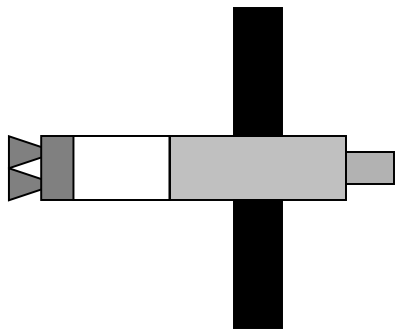
- Time 10 months
- Supplies for 36 month with failed recycling at minimal levels

Manned Mars Orbit Capture



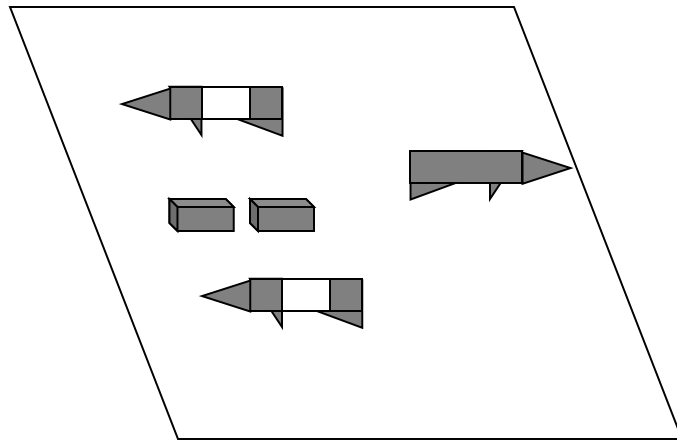
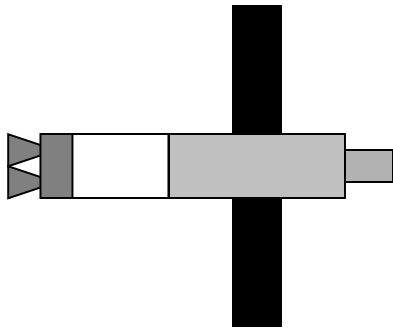
- 2/3 of Supplies Rem
- Initial C3 = 4 km/s
- Burn 1 Centaur
- DeltaV = 1.4 km/s
- Final orbit =
 - 3700 x 357000 km
 - (300x354000 alt)

Manned Mars Braking Decent



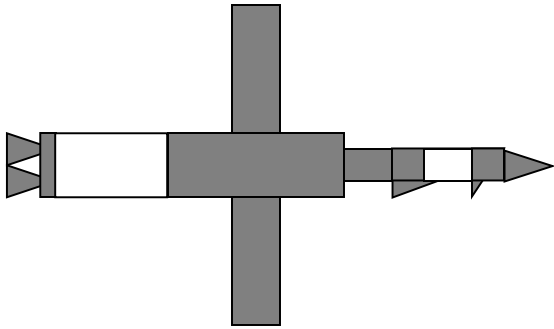
- Aerobrakes
- Land with beacon assistance
- $\Delta V = 200-500\text{m/s}$

Mars Surface Stay



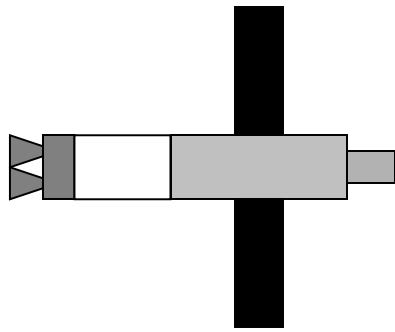
- Two+ early supply missions
- Surface stay 1.5 Yrs
- Supplies for 36 month with failed recycling at minimal levels
- Resupply option
- Abort to orbit early

Mars Orbit Rendezvous



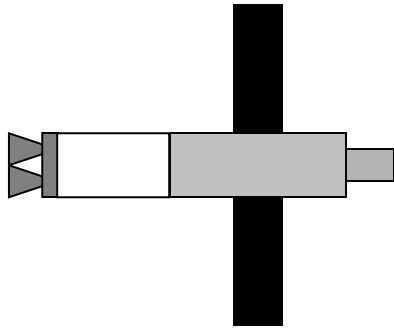
- Lander fueled for ascent to orbit
- final orbit matched with transit hab
- Offload extra supplies
- Hab may have lowered orbit during stay

Manned Trans Earth Impulse



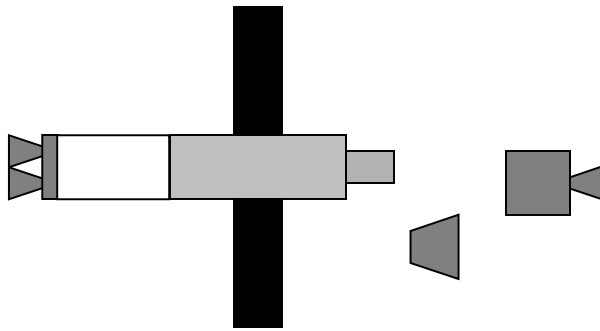
- Leave Lander
- Burn 1/2 of Centaur
- $\Delta V = 1.47$
- Final final C3=4km/s

Manned Earth Orbit Capture



- Initial C3 = 4.2 km/s
- Burn 0.125 of Centaur
- DeltaV = 0.8 km/s
- Final orbit =
 - 300x 787000 km
- Alternate RV
 - 3000kg vs 4200kg

Optional Crew Recovery



- Parking orbit
- Rendezvous with crew transfer vehicle
- Transit hab may be braked to Leo

System Costs

- Centaur Space Propulsion = \$20M
- Transit Habitat = \$30M
- Surface Habitat = \$30M
- Lander Development = \$200M
- Lander = \$30M
- Launch of 20000kg = \$20-300M

Supply Mission Cost

- 2 Centaurs \$40 M
- 1 Lander \$30 M
- Payload \$30 M
- Launch 3 units \$300M

– Total = \$400M

Manned Mission Cost

• 9 Centaurs	\$180 M
• 1 Lander	\$30 M
• Tranist Hab	\$30 M
• Launch 11 units	\$ 1100M
– Optimistic Total	= \$1340M

Total First Costs

- 3 Supply Missions \$1200M
 - 1 manned mission \$1340M
 - Lander development \$300M
- Total Cost for first mission = \$2840M
- +shuttle flights and assembly

Launch Costs

- 20 Launches
 - Assuming \$100M/Proton
 - Launch costs = \$2000M
- Total Price is \$2840M
- Reducing Launch to \$20M
 - \$1000kg (1/2x Zenit x2)
- Total Price is \$1240M